FTR - Flight Test Report Dieser Prüfbericht darf ohne schriftliche Zustimmung der EAPR nicht, auch nicht ar

Dieser Prüfbericht darf ohne schriftliche Zustimmung de	EAPR nicht, auch nicht auszugsweise, vervielfältigt v	verden.		
Manufacturer	independence gliders for real pilots	Type testing No.	EAPR-GS-0146/14	1=1
	Fly Market GmbH & Co.KG Am Schönebach 3 D-87637 Elsenberg	serial number	2k13-sample-M-092	Messer
Model	Cruiser 3 M	Lesstion	Achensee	E D-87730 Ba
		Location	Gardasee / Achensee	



Rev. 2.1 - 10.05.2013 EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany

Date of testing	20.01.2014	Minimum take off v 80 kg	veight	Maximum take off weight 105 kg			
Testpilot		Mike Küng		Mario Eder	0		
Harness		EAPR-Testequipment	E.	EAPR Testgurtzeug	S.		
Pilot's take off weight		80 kg	A CONTRACT	105 kg	A.		

Classification

A		



Test-criteria		Minimum take off weight			Maximum take o	off weight	Evalua
1. Inflation / take-off - 4.1.1							
Rising behavior		Smooth, easy	and constant rising	А	Smooth, easy and	d constant rising	A
Special take off technique required		No		A	No		A
2. Landing - 4.1.2							
Special landing technique required		No		А	No		А
3. Speeds in straight flight - 4.1.3							
Trim speed more than 30km/h		Yes		А	Yes		А
Speed range using the controls larger than 10km/	h	Yes		А	Yes		A
Minimum speed		Less than 25	km/h	A	Less than 25 km/h	h	A
4. Control movement - 4.1.4		•					
Max. weight in flight up to 80kg				-			-
Max. weight in flight 80 to 100kg				-			-
Max. weight in flight greater than 100kg		Increasing	>65 cm	А	Increasing	>65 cm	A
5. Pitch stability exiting accelerated flight - 4.1	.5						
Dive forward angle on exit		Dive forward I	less than 30°	A	Dive forward less	than 30°	A
Collapse occurs		No		А	No		A
6. Pitch stability operating controls during acce	elerated f	light - 4.1.6					
Collapse occurs		No		A	No		A
7. Roll stability and damping - 4.1.7							
Oscillations		Reducing		А	Reducing		А
8. Stability in gentle spirals - 4.1.8		•					
Tendency to return to straight flight		Spontaneous	exit	А	Spontaneous exit	А	
9. Behaviour in a steeply banked turn - 4.1.9		• •					
Sink rate after two turns		12m/s to 14m	/s	А	12m/s to 14m/s		А
10. Symmetric front collapse - 4.1.10							
Entry		Rocking back less than 45°		А	Rocking back less than 45°		А
Recovery	trim speed		in less than 3 sec	A	Spontaneous in less than 3 sec		A
Dive forward angle on exit	<u>ä</u>	0° - 30°	Keeping course	Α	0° - 30°	Keeping course	А
Cascade occurs	tr	No	<u>+</u> · · ·	A	No		A
Entry	σ	Rocking back less than 45°		A	Rocking back less than 45°		A
Recovery	accelerated	Spontaneous	in less than 3 sec	А	Spontaneous in le	ess than 3 sec	A
Dive forward angle on exit	cce	0° - 30°	Entering a turn of less than 90°	A	0° - 30°	Keeping course	A
	m	No		Α	No	-	А

Flight Test Report -

Deep stall achieved		Yes				Yes			
Recovery					А	Spontaneous in less than 3 sec			А
•		Spontaneous in less than 3 sec							
ive forward angle on exit hange of course		0° - 30° Changing course less than 45°			A	0° - 30° Changing course less than 45°			A
Cascade occurs		No			A	No			A
12. High angle of attack recovery - 4.1.12									
Recovery		Spontaneous in less than	n 3 sec		А	Spontaneous in	less than 3 sec		А
Cascade occurs		No			A	No			A
13. Recovery from a developed full stall - 4.1.13	3								
Dive forward angle on exit		0° - 30°			А	0° - 30°			А
Collapse Cascade occurs (other than collapse)		No collapse No			A	No collapse No			A
Rocking backward		Less than 45°			A	Less than 45°			A
Line tension		Most lines tight			A	Most lines tight			A
14. Asymmetric collapse (trim speed) - 4.1.14									
Change of course until re-inflation	ø	< 90° Dive or	roll angle	0° - 15°	А	< 90°	Dive or roll angle	0° - 15°	А
Re-inflation behavior	trim speed, max 50% collapse	Spontaneous re-inflation			А	Spontaneous re-	inflation		А
	trim speed, < 50% colla	•					innauon		
Total change of course Collapse on the opposite side occurs	rim 50%	Less than 360° No			A	Less than 360° No			A
Twist occurs	t max	No			A	No			A
Cascade occurs		No			A	No			A
Change of course until re-inflation	Se	90° - 180° Dive or	roll angle	0° - 15°	А	< 90°	Dive or roll angle	15° - 45°	А
Re-inflation behavior	trim speed, max 75% collapse	Spontaneous re-inflation			А	Spontaneous re-	inflation		А
Total change of course	spe 3% o	Less than 360°			A	Less than 360°			A
Collapse on the opposite side occurs	trim ax 755	No			A	No			A
Twist occurs	ma	No			Α	No			Α
Cascade occurs		No			A	No			A
Change of course until re-inflation	Se	90° - 180° Dive or	roll angle	0° - 15°	А	< 90°	Dive or roll angle	0° - 15°	А
Re-inflation behavior	accelerated, max 50% collapse	Spontaneous re-inflation			А	Spontaneous re-inflation			А
Total change of course	elera)% c	Less than 360°			A	Less than 360°			A
Collapse on the opposite side occurs	acce IX 50	No			A	No			A
Twist occurs	ma	No			A	No			A
Cascade occurs		No 90° - 180° Dive or	roll angle	0° - 15°	A	No < 90°	Dive or roll angle	15° - 45°	A
Change of course until re-inflation	accelerated, max 75% collapse		÷	0" - 15"				15" - 45"	A
Re-inflation behavior	accelerated, x 75% colla	Spontaneous re-inflation			A	Spontaneous re-inflation			A
Total change of course	cele 75%	Less than 360°		A	Less than 360° No			A	
Collapse on the opposite side occurs Twist occurs	ac max	No			A	No			A
Cascade occurs	_	No			А	No			А
15. Directional control with a maintained asymmetry	netric col								
Able to keep course straight		Yes			A	Yes			A
180° turn away from the collapsed side possible in 10 sec		Yes			A	Yes			A
Amount of control range between turn and stall or spin		More than 50% of the symmetric control travel			А	More than 50% of	f the symmetric o	ontrol travel	А
16. Trim speed spin tendency - 4.1.16									
Spin occurs		No			А	No			А
17. Low speed spin tendency - 4.1.17									
Spin occurs 18. Recovery from a developed spin - 4.1.18		No			A	No			A
Spin rotation angle after release		Stops spinning in less th	an 90°		A	Stops spinning in	1 less than 90°		A
Cascade occurs		No			A	No			A
19. B-line-stall - 4.1.19 Change of course before release		Changing course less the	an 45°		Α	Changing course	less than 45°		A
Behaviour before release					A				A
		Remains stable with straight span				Remains stable with straight span			
Recovery		Spontaneous in less than	n 3 sec		A	Spontaneous in less than 3 sec			A
Dive forward angle on exit Cascade occurs		0° - 30° No			A A	0° - 30° No			A
20. Big ears - 4.1.20									
Entry procedure		Special device required			А	Special device re	quired		А
Behaviour during big ears		Stable flight			A	Stable flight			A
Recovery		Spontaneous in less than	n 3 sec		А	Spontaneous in	less than 3 sec		А
Dive forward angle on exit		0° - 30°			A	0° bis 30°			A
21. Big Ears in accelerated flight - 4.1.21		<u> </u>			<u>_</u>				
		Special device required			А	Special device re	auired		А
Entry procedure Special device required					941104				
Behaviour during big ears		Stable flight			A				A
Recovery		Spontaneous in 3 to 5 se	9 C		A	Spontaneous in less than 3 sec			A
Dive forward angle on exit Behaviour immediately after releasing the accelara	tor while	0° - 30°			A	0° bis 30°			A
Behaviour immediately after releasing the accelarator while maintaining big ears		Stable flight			A	Stable flight			A

Tendency to return to straight flight	Spontaneous exit	A	Spontaneous exit		
Turn angle to recover normal flight	Less than 720°, spontaneous recovery	А	А		
23. Alternative means of directional control - 4.1.2	23		•		
180° turn achievable in 20 sec	Yes	А	Yes	А	
Stall or spin occurs	No	А	No	А	
24. Any other flight procedure and/or configuratio	n described in the user's manual - 4.1.24				
Procedure works as descibed		NA		NA	
Procedure suitable for novice pilots		NA		NA	
Cascade occurs		NA		NA	
25. Remarks of testpilot:					
Copyright Ralf Antz 2014	This Fligh		l was generated automatically and is valid wi	thout signature	